



Department of Transportation  
**Federal Aviation Administration**  
Aircraft Certification Service  
Washington, D.C.

**TSO-C5f**

Effective  
Date: {mm/dd/yy}

# Technical Standard Order

## (PROPOSED)

**Subject: DIRECTION INSTRUMENT, NON-MAGNETIC (GYROSCOPICALLY STABILIZED)**

1. **PURPOSE.** This technical standard order (TSO) is for manufacturers applying for a TSO authorization (TSOA) or letter of design approval (LODA). In it, we (the Federal Aviation Administration, or FAA) tell you what minimum performance standards (MPS) your, non-magnetic (gyroscopically stabilized), direction instruments must first meet for approval and identification with the applicable TSO marking.
2. **APPLICABILITY.** This TSO affects new applications submitted after its effective date.
  - a. All prior revisions to this TSO are no longer effective. Generally we will not accept applications after the effective date of this TSO. However, we may do so up to six months after it, if we know that you were working against the earlier MPS before the new change became effective.
  - b. Direction instruments, non-magnetic (gyroscopically stabilized), approved under a previous TSO authorization may still be manufactured under the provisions of their original approval.
  - c. Major design changes to non-magnetic (gyroscopically stabilized) direction instruments approved under previous versions of this TSO will require a new authorization under this TSO. See Title 14 of the Code of Federal Regulations (14 CFR) § 21.611(b).
3. **REQUIREMENTS.** New models of non-magnetic (gyroscopically stabilized) direction instruments identified and manufactured on or after the effective date of this TSO must meet the MPS in SAE International's Aerospace Standard (AS) 8021 *Minimum Performance Standard for Direction Instrument, Non-Magnetic (Gyroscopically Stabilized)*, dated March 16, 1981.
  - a. **Functionality.** This TSO's standards apply to equipment intended to provide directional guidance during flight.

**b. Failure Condition Classification.** Failure of the function defined in paragraphs 3 and 3.a of this TSO is a *major* failure condition. Develop the system to, at least, the design assurance level equal to this failure condition classification.

**c. Functional Qualification.** Demonstrate the required performance under the test conditions in AS 8021.

**d. Environmental Qualification.** Test the equipment according to RTCA, Inc. document RTCA/DO-160E, *Environmental Conditions and Test Procedures for Airborne Equipment*, dated December 9, 2004.

**e. Software Qualification.** If the article includes a digital computer, develop the software according to RTCA/DO-178B, *Software Considerations in Airborne Systems and Equipment Certification*, dated December 1, 1992.

**f. Electronic Hardware Qualification.** If the article includes a complex custom micro-coded component, develop the component to the guidance in FAA advisory circular (AC) 20-152, *RTCA, Inc. Document RTCA/DO-254, Design Assurance Guidance for Airborne Electronic Hardware*. The hardware design assurance level should be consistent with the failure condition classification defined in paragraph 3.b of this TSO.

**g. Deviations.** We have provisions for using alternate or equivalent means of compliance to the criteria in the MPS of this TSO. If you invoke these provisions, you must show that your equipment maintains an equivalent level of safety. Apply for a deviation under 14 CFR § 21.609 before submitting your data package.

#### **4. MARKING.**

**a.** Mark at least one major component permanently and legibly with all the information in 14 CFR § 21.607(d), except for the following:

(1) 14 CFR § 21.607(d)(2). Use the name, type, and part number. Do not use the optional model number; and

(2) 14 CFR § 21.607(d)(3). Use the date of manufacture instead of the optional serial number.

**b.** Also, mark the following permanently and legibly, with at least the manufacturer's name, subassembly part number, and the TSO number:

(1) Each component that is easily removable (without hand tools),

(2) Each interchangeable element, and

(3) Each subassembly of the article that you determined may be interchangeable.

c. If the component includes a digital computer, then the part number must include hardware and software identification. Or, you can use a separate part number for hardware and software. Either way, you must include a means to show the modification status.

**NOTE:** Similar software versions, approved to different software levels, must be differentiated by part number.

d. Identify deviations granted to the article by marking “Deviation. See installation/instruction manual (IM)” after the TSO number. You can abbreviate the marking to “Dev. See IM.”

e. When applicable, identify the equipment as an incomplete system or state that the appliance performs functions beyond those described in paragraphs 3 and 3.a of this TSO.

**5. APPLICATION DATA REQUIREMENTS.** As a TSO manufacturer-applicant, you must give the FAA’s aircraft certification office (ACO) manager responsible for your facilities a statement of conformance, as specified in 14 CFR § 21.605(a)(1) and one copy each of the following technical data to support our design and production approval. (Under 14 CFR § 21.617(a)(2), LODA applicants submit the same data through their civil aviation authority:)

a. Operating instructions and equipment limitations in an IM, sufficient to describe the equipment’s operational capability. Describe any deviations in detail. If needed, identify equipment by part number, version, revision, and criticality level of software/hardware, classification for use, and environmental categories.

b. Installation procedures and limitations in an IM, sufficient to ensure that the non-magnetic (gyroscopically stabilized) direction instrument, when installed according to the installation procedures, still meets this TSO’s requirements. Limitations must identify any unique aspects of the installation. Finally, the limitations must include a note with the following statement:

The conditions and tests for TSO approval of this article are minimum performance standards. Those installing this article, on or in a specific type or class of aircraft, must determine that the aircraft installation conditions are within the TSO standards. TSO articles must have separate approval for installation in an aircraft. The article may be installed only according to 14 CFR part 43 or the applicable airworthiness requirements.

c. Schematic drawings of the installation procedures.

d. Wiring diagrams of the installation procedures.

- e. List of components, by part number, that make up the non-magnetic (gyroscopically stabilized) direction instrument complying with the standards in this TSO. Include vendor part number cross-references, when applicable.
- f. A component maintenance manual (CMM), covering periodic maintenance, calibration, and repair, for the continued airworthiness of installed non-magnetic (gyroscopically stabilized) direction instruments. Instructions should include recommended inspection intervals and service life. Describe the details of deviations granted, as noted in paragraph 5.a of this TSO.
- g. Material and process specifications list.
- h. The quality control system (QCS) description required by 14 CFR §§ 21.143 and 21.605(a)(3) including functional test specifications. The QCS should ensure that you will detect any change to the equipment that could adversely affect compliance with the TSO MPS, and reject the item accordingly. (Not required for LODA applicants.)
- i. Manufacturer's TSO qualification test report.
- j. Nameplate drawing with the information required by paragraph 4 of this TSO.
- k. List of all drawings and processes (including revision level), that define the article's design. For a minor change, follow the directions in 14 CFR § 21.611(a). Show any revisions to the drawing list only on our request.
- l. An environmental qualifications form as described in the environmental qualifications document referenced in paragraph 3.d of this TSO for each component of the system.
- m. If the article includes a digital computer: a plan for software aspects of certification (PSAC), software configuration index, and software accomplishment summary. We recommend that you submit the PSAC early in the software development process. Early submittal allows us to quickly resolve issues, such as partitioning and determining software levels.
- n. If the article includes a complex custom micro-coded component: a plan for hardware aspects of certification (PHAC), hardware verification plan, top-level drawing, and hardware accomplishment summary. We recommend that you submit the PHAC early in the software development process. Early submittal allows us to quickly resolve issues.

**6. MANUFACTURER DATA REQUIREMENTS.** Besides the data given directly to the FAA, have the following technical data available for review by the responsible ACO or civil aviation authority:

- a. The functional qualification specifications for qualifying each production article to ensure compliance with this TSO.
- b. Equipment calibration procedures.

- c. Corrective maintenance procedures within 12 months after TSO authorization.
- d. Schematic drawings.
- e. Wiring diagrams.
- f. Material and process specifications.
- g. The results of the environmental qualification tests conducted per paragraph 3.d. of this TSO.
- h. If the article includes a digital computer, the appropriate documentation defined in RTCA/DO-178B or the most current revision, including all data supporting the applicable objectives in Annex A, *Process Objectives and Outputs by Software Level*.
- i. If the article includes a complex micro-coded component, the appropriate hardware life cycle data in combination with design assurance level, as defined in RTCA/DO-254, Appendix A, Table A-1.

**7. FURNISHED DATA REQUIREMENTS.** If furnishing one or more articles to one entity (such as an operator or repair station), provide the following items for each article manufactured under this TSO:

- a. One copy of the data in paragraphs 5.a through 5.g of this TSO. Add any other data needed for the proper installation, certification, use, or continued airworthiness of the non-magnetic (gyroscopically stabilized), direction instrument.
- b. If the article performs functions beyond those described in paragraphs 3 and 3.a of this TSO, send one copy of the data in paragraphs 5.i through 5.n.

**8. HOW TO GET REFERENCED DOCUMENTS.**

- a. Order SAE documents from SAE International, 400 Commonwealth Drive, Warrendale, PA 15096-0001. Telephone (724) 776-4970, fax (724) 776-0790. You can also order copies through the SAE website at [www.sae.org](http://www.sae.org).
- b. Order RTCA documents from RTCA Inc., 1828 L Street NW, Suite 805, Washington, D.C. 20036. Telephone (202) 833-0330, fax (202) 833-9434. You can also order copies on the RTCA website at [www.rtca.org](http://www.rtca.org).
- c. Order copies of 14 CFR parts from the Superintendent of Documents, Government Printing Office, P.O. Box 37154, Pittsburgh, PA 15250-7954. You can also order copies through

the Government Printing Office website at [www.access.gpo.gov](http://www.access.gpo.gov). Select “Access,” then “Online Bookstore.” Select “Aviation,” then scroll to “Code of Federal Regulations.”

**d.** You can find a current list of technical standard orders and advisory circulars on the FAA Internet website Regulatory and Guidance Library at [www.airweb.faa.gov/rgl](http://www.airweb.faa.gov/rgl). You will also find the TSO Index of Articles at the same site.

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